



GLAST Monthly PSR-December 2003

System Engineering

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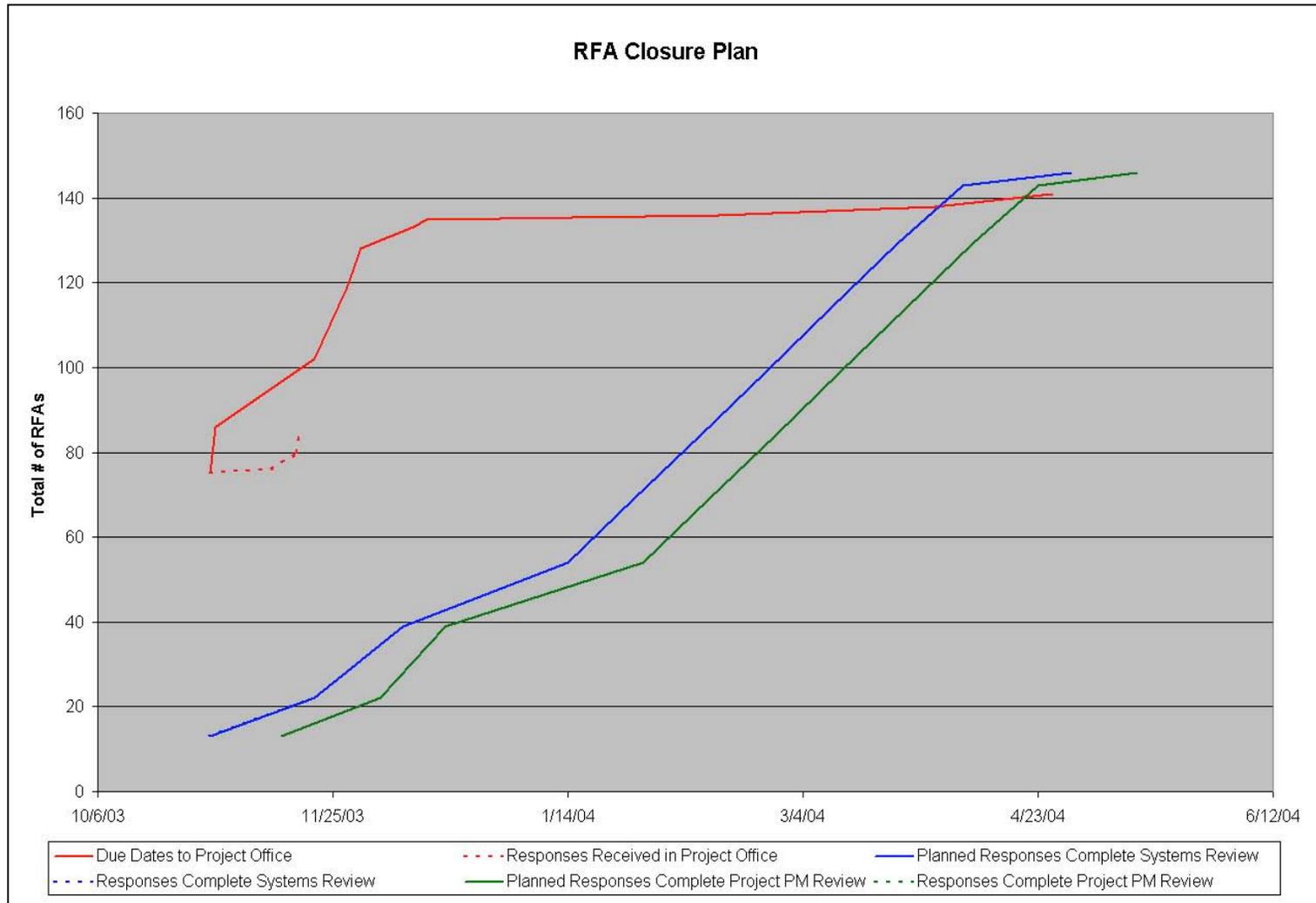
RFA Status

- Reviewed 29 RFA responses with systems & discipline engineers on 11/21, 19 approved, 10 needed additional review or updates based on comments received
- 13 RFA responses sent to PM for review on 10/16, awaiting final approval before sending to Originators, 19 additional responses sent to PM for review on 12/3
- Received 3 additional RFA responses from SLAC this month
- Awaiting updated responses from Spectrum based on previous comments
- Reviewed the open LAT PDR & delta PDR RFAs and assigned actions to close each
- Generated a closure plan for receipt and approval of all RFA responses, already behind Spectrum, SLAC, and Project have missed submittal dates
- RFA tracking tool now on-line on the NGIN website under “Engineering Files”. Only Project Office access to these files

Review	Total # of RFAs	Systems Review Status		Project Review Status		Code 300 Status	Notes
		In Review	Complete	In Review	Complete	# Closed	
Mission SRR	22		22		22	22	All Closed
LAT PDR	57	9	48	5	43	43	All Closed
GBM PDR & FSW PDR	27	4	23		23	23	4 Open, 1 Withdrawn
LAT delta PDR	20	3	17		17	17	3 Open
SC PDR	43	10	10	10			All Open
SC FSW PDR	14	5	1	1			All Open
LAT CDR	37	20	2	2			All Open
MPDR	14	3	6	6			All Open
GS SRR	17	4	8	8			All Open
Totals	251	58	137	32	105	105	



RFA Closure Plan





GLAST Project CCR Status



▶ **No CCBs held in November**

- *2 deferred CCRs (0184 & 0185, SIIS/SDIS) were updated per the CCB action and approved.*
- *1 additional action to resubmit the Digital Photos CCR as R1*

▶ **Currently there are 18 CCRs in the System**

- *Discussing EMI Requirements today (CCR 0205), SAI needs SSR EMI requirements by 12/12 or cost/schedule impact; if signed out of board, we should also review CCR 0206, the EMI CDRL delivery dates*
- *Plan to schedule next CCB for week of 12/15; agenda to include Rebaseline GSRD & Ops Con, SSMAP, Slews per Orbit, TVAC Dwell Time, LAT Quasi-Static Limit Load Factors, LAT-SC Radiator Interface Pad Size, SEMP updates for Lessons Learned and Metric Units and CDRL 1 updates*
- *Still need SAI response for Increasing SC Power Capability to LAT*
 - *Spectrum position is they can modify solar array later, so why force them to make a decision now*
- *Working with SAI on comments to the ITAR document*



Independent Safe Mode Processor Status



▶ ***Independent Safe Mode Processor (ISMP) Status***

- *Stop work order issued for Independent Safe Mode CSCI (software load part of mission computer)*
- *ISMP Special Study # initiated - concern is lack of RAD750 flight heritage*
- *Separate and distinct processor from RAD750 for safe mode applications*
- *Three requirements iterations in last 3 weeks between project and Spectrum*
- *“Final” Special Study requirements set available to be delivered*

▶ ***Spectrum Astro Special Study short term schedule***

- *12/5 Requirements due from GPO (REQUIREMENTS FROZEN)*
- *12/12 Feedback discussion/Trades Discussion (DESIGN FROZEN)*
- *12/29 Discuss Midterm results*
- *Project desires face-to-face in mid December*

▶ ***Requirements with significant cost impact***

- *Communications / Data Storage – Storage and downlink of data critical for diagnostics and recovery operations*
- *Simultaneous processor operations and re-programmability*
- *1553 data bus interface (for SIRU) – need for SIRU tied to communications management operations plan*



Accomplishments



- ▶ ***Project proceeding with 5 Mil Solar Array Kapton***
 - *NTE ROM from Study #9 overdue*
- ▶ ***Burst Alert Latency estimates being refined***
 - *Current estimate is ~7-9 seconds*
 - *Will discuss with S. Ritz and N. Gerhls to determine if acceptable*
- ▶ ***GLAST Orbital Debris Assessment Report on hold***
 - *Awaiting meeting with Deputy AA Code S to determine if removal of propulsion system is viable from a Code S perspective*
 - *If viable (removal of prop system viable), next step to meet jointly with Code Q*
 - *If not, thanks for all the hard work*
 - *ODA Report will be modified accordingly and sent to HQ*



Observatory STOP Analysis Status



- ▶ **Cycle 1: Unit Thermal Gradient Analysis (UTGA) w/ delta-PDR Models**
 - *SAI completed unit thermal gradient analysis using delta-PDR models with updated thermal properties (CTEs) and LAT boresight definitions*
 - *Thermal/mechanical distortions < 6 arc-sec per 1 deg C in all axes (X,Y,Z)*
- ▶ **Cycle 2: UTGA w/ interim LAT CDR models and Four Static Thermal Case Analysis**
 - *SAI to perform unit thermal gradient analysis using updated models*
 - *11/15 - SAI integrated LAT & SC models and perform model checkout*
 - *12/08 – SAI to perform unit thermal gradient analysis (compare to Cycle 1 results)*
 - *SAI to perform static case runs using integrated observatory models*
 - *11/25 – SAI completed initial run with four static thermal cases; provide results to GPO*
 - *12/05 – GPO-Thermal and Swales refine mapping and provide four static load cases to SAI*
 - *01/07 – SAI to complete four static case runs*
 - *01/21 – Spectrum to provide static thermal case analysis results*
- ▶ **Cycle 3: Observatory-Level STOP Analysis using final LAT CDR models**
 - *Transient case analysis using worst-case on-orbit thermal cases identified and Cycle 1&2 results*
 - *12/05 – SLAC to deliver LAT CDR FEM (10.07) to GPO*
 - *GPO considering having Swales (P.Baird) perform Cycle 3 analysis*
- ▶ **Cycle 4: Reperform Cycle 3 using T/V-correlated LAT models**
- ▶ **Cycle 5: Reperform Cycle 3 using T/V-correlated observatory models**



Mass Budget



	Mass (kg)				
	Allocation	Estimate	Margin	%	Delta
▶ Dry SC	1169	912	257	28	6
▶ SC including propellant	1530	1273	258	20	6
▶ LAT	3000	2756	244	9	6
▶ GBM	<u>97</u>	<u>84</u>	<u>13</u>	<u>15</u>	<u>0</u>
▶ Observatory mass	4627	4113	515	13	12

- ▶ **Note: Positive Delta represents a decrease in margin from last month**
- ▶ **Estimate includes 39 kg LAT Mass increase expected from X-LAT Plate Peer Review**
- ▶ **Delta II Heavy throw weight to 575 km with cg at 1.37 m = 4627 kg**
- ▶ **70% of LAT mass estimate is measured**
- ▶ **LAT has entered the CDR level of maturity, and is showing 9% margin.**
- ▶ **AIAA recommendations for the mass of flight systems recommend holding 7.2% margin at the PDR stage of the LAT program (198kg), and the LAT is holding 1.9% reserve on top of that (55kg).**
- ▶ **LAT is carrying 29% margin on the unmeasured LAT mass of 827kg**

12-04-2003



Power Budget



Orbit Average Power (Watts)

	<i>Allocation</i>	<i>Estimate</i>	<i>Margin</i>	<i>%</i>	<i>Delta</i>
Spacecraft	985	734	251	34	4
LAT	650	573	77	13	-15
GBM	<u>65</u>	<u>55</u>	<u>10</u>	<u>18</u>	<u>0</u>
Observatory total	1700	1362	338	25	-11

Note: Negative Delta represents an increase in margin from last month

12-04-2003



Communication Link Margins



<i>Link</i>	<i>Type</i>	<i>Data Rate (kbps)</i>	<i>Coverage</i>	<i>Link Margin (dB)</i>
<i>Ku Band Science</i>	<i>Data, TLM</i>	<i>40,000</i>	<i>> = 5 min per orbit</i>	<i>4.9</i>
<i>GN</i>	<i>CMD</i>	<i>2</i>	<i>90% Spherical Coverage</i>	<i>39.0</i>
<i>GN</i>	<i>TLM</i>	<i>2,500</i>	<i>90% Spherical Coverage</i>	<i>10.5</i>
<i>TDRSS MA</i>	<i>CMD</i>	<i>0.250</i>	<i>90% Spherical Coverage</i>	<i>1.4 (-.8)</i>
<i>TDRSS SA</i>	<i>CMD</i>	<i>4</i>	<i>90% Spherical Coverage</i>	<i>4.8 (+.1)</i>
<i>TDRSS MA</i>	<i>TLM</i>	<i>1</i>	<i>90% Spherical Coverage</i>	<i>1.1 (-.8)</i>
<i>TDRSS SA</i>	<i>TLM</i>	<i>1</i>	<i>90% Spherical Coverage</i>	<i>10.7</i>

- Ku Band Link Margin introduce for first time
- TDRSS estimates adjusted to incorporate conservatism in GSFC CLASS TDRSS analysis

12-04-2003



Top Issues/Concerns

- ▶ ***LAT data rate increase CCR needs quick go-ahead for Spectrum to purchase additional 32 M 64 M of memory for SSR***
 - *Allows 21/27 hours of science data storage at LAT 1.2 Mbps avg rate*
 - *Approval would help wrt SSR CDR in January*
- ▶ ***Held discussions on current monitoring***
 - *Spectrum agreed that some additional capability would help*
 - *Need to agree on specifics soon*
- ▶ ***Power on at launch configuration***
 - *What spacecraft components does the Project consider need to be on at launch?*
 - *C&DH in order to monitor spacecraft during ascent and separation*
 - *Transmitter(s) to provide telemetry in real-time*
 - *SSR to record (?)*
 - *GNC components necessary for sun capture (?)*
 - *Precise configuration definition needed as it may impact component (as well as system) specs and testing*